





Research & Development Project

## **HISTORY AND SKILLS**

Research & Development of "Small UAV with electrically powered propeller"

#### **Description of the solution:**

- Airframe, electronics, 2 battery sets
- 1 spare Airframe, battery charger
- Transport hardtop casing and backpack for field operation
- Interface touchscreen laptop and video antenna

UAV – Small Unmanned Reconnaissance Aerial Vehicle (2006)



Research & Development Project

## **HISTORY AND SKILLS**

#### Layout of main drive and control units





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## **HISTORY AND SKILLS**



#### **Technical parameters:**

- Span: 220 cm
- Lenght: 125 cm
- Weight: cca 3 kg
- Payload : up to 1 kg
- Silent electric motor
- Endurance : up to 2 hours
- Range : 50 km

Real flight tests during dusk and night



Research & Development Project

Current R&D project: UAV - Model MS

### Project target: 5

Development of middle Unmanned Reconnaissance Aerial Vehicle equipped with a Turbojet engine - Model MS

The UAV is designed for subsonic speeds (up to 400km/h), middle sized, intended for reconnaissance, surveilance or special applications. Flight control system will be Auto, Semi and Manual.





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Actual R&D project: UAV - Model MS



#### Aircraft parameters:

- Airframe configuration classical
- V-tail type
- Length L=4 m
- Airframe body diameter D=0,7 m
- Wing span W=5 m
- Chord span B=0,6 m
- Takeoff weight 300 kg

- Payload weight 30 kg
- Engine type turbojet engine TJ100
- Control surfaces elevator , rudder
- Kill switch (cut off)
- Aircraft material combination of graphite composites and aluminum alloy



Research & Development Project

## Actual R&D project: UAV - Model MS

#### **AUTOPILOT – Flight Control System (FCS)**



#### Proposed configuration of the UAV provide

- Great flight stability
- High efficiency of the thrust engine
- Small aerodynamic resistance



Research & Development Project

Actual R&D project: UAV - Model MS



![](_page_8_Picture_0.jpeg)

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Current R&D project: UAV - Model MS

![](_page_8_Figure_4.jpeg)

#### The autopilot main parameters:

- Microcomputer control unit
- GPS (GLONASS)
- 3-axis MODE (MEMS gyro, MEMS accelerometer)
- 3-axis magnetic sensor, an accurate barometric pressure sensor
- The UAV position control algorithm
- The frequency of the internal management of UAV
- Position in space 100 Hz,
- The navigation signals frequency of UAE– 10 Hz
- Controlled elements: rudder, elevator, ailerons

Navigation modes: controlled flight from the ground control station and the flight controlled by an UAV video sensor.

![](_page_9_Picture_0.jpeg)

Research & Development Project

Current R&D project: UAV - Model MS

#### Telemetry data transmitted to the Ground Control Station and Communication System

- Supply and fuel consumption (voltage and battery current consumption)
- Received signal GPS (GLONASS)
- Position of UAV in the space
- Parameters for a virtual dashboard for UAV Ground Control Station
- Data from the payload (payloads data)
- Downlink Bandwidth 10 Mb/s
- Uplink Bandwidth 1 Mb/s
- Frequency Band 5150 5250 MHz

![](_page_9_Picture_13.jpeg)

![](_page_10_Picture_0.jpeg)

Research & Development Project

## Current R&D project: UAV - Model MS

#### Flight Control System installation requirements

- Aircraft shall be equipped with power supply for the FCS (10-30 DC, 300 mA) and for servos
- Aircraft servos shall be controlled by PWM (1-2 msec, 50 Hz)
- Flight control system shall be istalled as close as possible to the center of the airframe gravity
- Data exchange with radio modem and turning antenna should be done trough
- Ground Station shall be inbuilt to the computer with installed OS Windows

![](_page_10_Picture_10.jpeg)

![](_page_11_Picture_0.jpeg)

Research & Development Project

## Current R&D project: UAV - Model MS

#### Aircraft takeoff and landing

Takeoff is secured by mechanical or pneumatical catapult, which shall be mounted on truck trailer.

Landing is secured by parachute system.

![](_page_11_Picture_7.jpeg)

![](_page_12_Picture_0.jpeg)

Research & Development Project

Current R&D project: UAV - Model HS

### Project target: 5

Development of middle Unmanned Reconnaissance Aerial Vehicle equipped with a Turbojet engine - Model HS

The UAV is designed for transsonic speeds (up to 800km/h at altitude of 8000m), middle sized, intended for reconnaissance, surveilance or special applications. Flight control system will be Auto, Semi and Manual.

![](_page_12_Picture_7.jpeg)

![](_page_13_Picture_0.jpeg)

Research & Development Project

Current R&D project: UAV - Model HS

![](_page_13_Picture_4.jpeg)

#### Aircraft parameters:

- Airframe configuration classical
- V-tail type
- Length L=4 m
- Airframe body diameter D=0,7 m
- Wing span W=3 m
- Chord span B=0,6 m
- Takeoff weight 300 kg

- Payload weight 30 kg
- Engine type turbojet engine TJ100
- Control surfaces elevator , rudder
- Kill switch (cut off)
- Aircraft material combination of graphite composites and aluminum alloy

Most parameters of UAV Models MS and HS are identical.

![](_page_14_Picture_0.jpeg)

Research & Development Project

Current R&D project: UAV - Model HS

![](_page_14_Figure_4.jpeg)

![](_page_15_Picture_0.jpeg)

Research & Development Project

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# Current R&D project: UAV - Model HS

#### **Visualization gallery**

![](_page_15_Picture_5.jpeg)

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#### **Possible future cooperation on UAV development:**

- DSSI a.s. should develop and build for customer an UAV MS or Model HS on turnkey
- DSSI a.s. is open to cooperate on R&D of new UAV models with potential partners based on contract.

![](_page_16_Picture_5.jpeg)

#### DSSI a.s.

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